CE 203 - Term 121
Solution of HOMEWORK NO. 3
1- Solve problem F2-2 in the textbook (p. 74) using the given revised data: angle of rotation is 0.015 degrees (instead of .02).

Solution I:
The figure shows the orientation of the rigid arm $A B C$ after rotation. From the figure for member EC:

$$
\begin{aligned}
& \alpha=\tan ^{-1}\left(\frac{1200 \mathrm{~mm}}{600 \mathrm{~mm}}\right)=63.435^{\circ} \\
& \sigma=90-\alpha+0.015=26.58^{\circ} \\
& L_{A E}=\sqrt{(600 \mathrm{~mm})^{2}+(1200 \mathrm{~mm})^{2}}=1341.641 \mathrm{~mm}
\end{aligned}
$$

Applying the law of cosines:


$$
\begin{aligned}
L_{C E} & =\sqrt{\left(L_{A E}\right)^{2}+\left(L_{A C^{\prime}}\right)^{2}-2\left(L_{A E}\right) \cdot\left(L_{A C^{\prime}}\right) \cos \left(26.58^{\circ}\right)} \\
& =\sqrt{(1341.641 \mathrm{~mm})^{2}+(1200 \mathrm{~mm})^{2}-2(1341.641 \mathrm{~mm}) \cdot(1200 \mathrm{~mm}) \cdot \cos \left(26.58^{\circ}\right)} \\
& =600.3132 \mathrm{~mm}
\end{aligned}
$$

Normal strain formember $C E, \epsilon_{C_{E}}=\frac{L_{C E}-L_{C E}}{L_{C E}}=\frac{600.3132 \mathrm{~mm}-600 \mathrm{~mm}}{600 \mathrm{~mm}}$
For member $D B$ :

$$
\begin{aligned}
& B=\tan ^{-1}\left(\frac{600 \mathrm{~mm}}{400 \mathrm{~mm}}\right)=56.31^{\circ} \\
& \phi=90-\beta+0.015=33.705^{\circ} \\
& L_{A D}=\sqrt{(400 \mathrm{~mm})^{2}+(600 \mathrm{~mm})^{2}}=721.110 \mathrm{~mm} \\
& L_{B^{\prime} D}=\sqrt{(721.11 \mathrm{~mm})^{2}+(600 \mathrm{~mm})^{2}-2(721.11 \mathrm{~mm})(600 \mathrm{~mm}) \cdot \cos \left(33.705^{\circ}\right)}=400.156 \mathrm{~mm} \\
& =400.156 \mathrm{~mm} . \\
& \text { Normal strain for member BD, } \epsilon_{B D}=\frac{L_{B_{D}}-L_{B D}}{L_{B D}}=\frac{400.156 \mathrm{~mm}-400 \mathrm{~mm}}{400 \mathrm{~mm}} \\
&
\end{aligned}
$$

Problem 1 ( $\operatorname{con}^{2} t$ ).
Solution II:
For small strains, the same results can be obtained by approximating the elongation of both wires,
For member CE:

$$
\begin{aligned}
\Delta L_{C E}=\theta L_{A C} & =\left[\left(\frac{0.15^{\circ}}{180^{\circ}}\right) \cdot(\pi \mathrm{rad})\right] \cdot(1200 \mathrm{~mm})=0.31416 \mathrm{~mm} \\
\Rightarrow & \epsilon_{C E}=\frac{\Delta L_{C E}}{L_{C E}}=\frac{0.31416 \mathrm{~mm}}{600 \mathrm{~mm}}=5.236 \times 10^{-4} \mathrm{~mm} / \mathrm{mm}
\end{aligned}
$$

For member BD:

$$
\begin{aligned}
& \text { mber } B D: \\
& \Delta_{B D}=\left[\left(\frac{0.015^{\circ}}{180^{\circ}}\right) \cdot(\pi \mathrm{rad})\right] \cdot(600 \mathrm{~mm})=0.157 \mathrm{~mm} \\
& \Rightarrow \epsilon_{B D}=\frac{\Delta L_{B D}}{L_{B D}}=\frac{0.157 \mathrm{~mm}}{400 \mathrm{~mm}}=3.927 \times 10^{-4} \mathrm{~mm} / \mathrm{mm}
\end{aligned}
$$

2- Using the strain results of Problem (1) above, determine the tension in each cable given that for each cable: cross sectional area $=20 \mathrm{~mm}^{2} \& E=20 \mathrm{GPa}$. (Hint : you do not need equilibrium equations in this case)

Solution:
Using Hooke's Law, $\sigma=E \cdot \epsilon$

- For cable (wire) CE:

$$
\begin{aligned}
& \sigma_{C E}=E \cdot \epsilon_{C E}=\left(20 \times 10^{9} \mathrm{~Pa}\right) \cdot\left(5.22 \times 10^{-4} \frac{\mathrm{~mm}}{\mathrm{~mm}}\right) \\
& =10.44 \times 10^{6} \mathrm{~Pa}=10.44 \mathrm{MPa} \text {. } \\
& \therefore F_{C E}=G_{C E} \cdot A_{C E} \equiv\left(10.44 \times 10^{6} \mathrm{~Pa}\right) \cdot\left(20 \times 10^{-6} \mathrm{~m}^{2}\right) \\
& =208.8 \mathrm{~N} \text {. (Tension Force). }
\end{aligned}
$$

- For cable (wive) BD:

$$
\begin{aligned}
\sigma_{B D}=E_{B D} & =\left(20 \times 10^{9} \mathrm{~Pa}\right) \cdot\left(3.90 \times 10^{-4} \frac{\mathrm{~mm}}{\mathrm{~mm}}\right) \\
& =7.8 \times 10^{6} \mathrm{~Pa}=7.8 \mathrm{MPa} \\
F_{B D}=\sigma_{B D} \cdot A_{B D} & =\left(7.8 \times 10^{6} \mathrm{~Pa}\right) \cdot\left(20 \times 10^{-6} \mathrm{~m}^{2}\right) \\
& =156.0 \mathrm{~N} \cdot \text { (Tension Force). }
\end{aligned}
$$

3- Using the table given in problem 3-1 in your textbook (p. 98), plot the stress-strain diagram for the material accurately and to scale. Use the following revised data : diameter $=180 \mathrm{~mm}$ (instead of 150) and the length $=350 \mathrm{~mm}$ (instead of 300). Use the plot to determine the following properties : Modulus of Elasticity , the proportional limit, the ultimate stress , the failure stress, and the modulus of toughness.

## Solution:

The stress strain values of a concrete cylinder were calculated and tabulated in the next page using the following equations:

$$
\begin{gathered}
\text { Sress, } \sigma=\frac{\text { Load }}{\text { Area }}=\operatorname{load} /\left(\pi(0.09 \mathrm{~m})^{2}\right) \\
\text { Strain, } \varepsilon=\frac{\text { Contraction }(\mathrm{mm})}{\text { Gage Length }(\mathrm{mm})}=\frac{\text { Contraction }(\mathrm{mm})}{350 \mathrm{~mm}}
\end{gathered}
$$

- The plotted stress strain diagram is shown in the next page, and it represents the initial portion of a typical stress strain diagram of concrete.

Note: the ultimate strength of typical concrete is 35 MPa with the corresponding strain of $\mathbf{0 . 0 0 2} \mathbf{~ m m} / \mathrm{mm}$.

- The modulus of elasticity, $E$ (approximately) $=\frac{6.25 \mathrm{MPa}-0}{0.0003 \mathrm{~mm} / \mathrm{mm}-0}=20.833 \mathrm{GPa}$
- The proportional limit: it is not easy to determine.
- The maximum stress from the diagram $=\mathbf{1 0 . 4 1 3 8} \mathbf{~ M P a}$.
$\circ$ Both the ultimate stress $\left(\sigma_{u}\right)$ and the failure stress $\left(\sigma_{f}\right)$ cannot be obtained since the complete diagram up to failure is not shown .
- For the given portion of the diagram,
the modulus of toughness, $U_{t}$ (approximately) $=$ the entire area under the curve

$$
\begin{aligned}
& =(0.5) \times\left(10.4138 \times 10^{6}\right) \times(0.0005286 \mathrm{~mm} / \mathrm{mm}) \\
& =0.00275 \mathrm{MJ} / \mathrm{m}^{3}
\end{aligned}
$$

## Stress Strain Data

| Load <br> $(\mathbf{k N})$ | Stress <br> $(\mathbf{M P a})$ | Contraction <br> $(\mathrm{mm})$ | Strain <br> $(\mathrm{mm} / \mathrm{mm})$ |
| :---: | :---: | :---: | :---: |
| 0.0 | 0.0000 | 0.0000 | 0.0000000 |
| 25.0 | 0.9824 | 0.0150 | 0.0000429 |
| 47.5 | 1.8666 | 0.0300 | 0.0000857 |
| 82.5 | 3.2420 | 0.0500 | 0.0001429 |
| 102.5 | 4.0280 | 0.0650 | 0.0001857 |
| 127.5 | 5.0104 | 0.0850 | 0.0002429 |
| 150.0 | 5.8946 | 0.1000 | 0.0002857 |
| 172.5 | 6.7788 | 0.1125 | 0.0003214 |
| 192.5 | 7.5648 | 0.1250 | 0.0003571 |
| 232.5 | 9.1367 | 0.1550 | 0.0004429 |
| 250.0 | 9.8244 | 0.1750 | 0.0005000 |
| 265.0 | 10.4138 | 0.1850 | 0.0005286 |

Sress-Strain Diagram


4- The given solid block has the initial dimensions shown in the figure. Determine the final dimensions of the block, and the change in volume of the block. Given: $\mathrm{E}=80 \mathrm{GPa}, v=0.35$. (Note : the loads are applied on rigid thin plates only to distribute the stress over the area)

Solution:
The compression stress acting on the block along $y$-axis

$$
\begin{aligned}
\therefore \sigma_{y}= & \frac{-20 \mathrm{kN}}{A_{\text {thinplate }}} \\
= & \frac{-20 \times 10^{3} \mathrm{~N}}{(0.05 \mathrm{~m}) \cdot(0.06 \mathrm{~m})} \\
& -6.667 \times 10^{6} \mathrm{~Pa} \quad \text { (c) }
\end{aligned}
$$



Using Hookers law $\Rightarrow \epsilon_{y}=\frac{\sigma_{y}}{E}=\frac{-6.667 \times 10^{6} \mathrm{~Pa}}{80 \times 10^{9} \mathrm{~Pa}}=-83.338 \times 10^{-6} \mathrm{~m}$

$$
\Rightarrow \delta_{y}=\epsilon_{y} \cdot L_{y}=\left(-83.338 \times 10^{-6} \frac{\mathrm{~mm}}{\mathrm{~mm}}\right) \cdot(220 \mathrm{~mm})=-0.018334 \mathrm{~mm}
$$

$\Rightarrow$ Final dimension along $y=L_{y}+\delta_{y}=219.98167 \mathrm{~mm}$ Using the Poisson's ratio, $Z=0.35=-\frac{E_{\text {lat }}}{E_{\text {long }}}$

$$
\begin{aligned}
\Rightarrow \epsilon_{x}=\epsilon_{z}=-\nu \cdot \epsilon_{y} & =-(0.35) \cdot\left(-83.338 \times 10^{-6} \frac{\mathrm{~mm}}{\mathrm{~mm}}\right) \\
& =29.168 \times 10^{-6} \frac{\mathrm{~mm}}{\mathrm{~mm}} \\
\Rightarrow \delta_{x}=\epsilon_{x} \cdot L_{x} & =\left(29.168 \times 10^{-6} \frac{\mathrm{~mm}}{\mathrm{~mm}}\right) \cdot(50 \mathrm{~mm})=+0.001458 \mathrm{~mm} \\
\Rightarrow \delta_{z}=\epsilon_{z} \cdot L_{z} & =\left(29.168 \times 10^{-6} \frac{\mathrm{~mm}}{\mathrm{~mm}}\right) \cdot(60 \mathrm{~mm})=+0.00175 \mathrm{~mm}
\end{aligned}
$$

Final dimension along $x$-axis $=L_{x}+\delta_{x}=50.001458 \mathrm{~mm}$.

$$
\sim \sim \sim \sim z \text {-axis }=L_{z}+\delta_{z}=60.00175 \mathrm{~mm}
$$

$\therefore$ The change in volume, $\Delta V=$ The final volume - initial volume

$$
\begin{aligned}
& =(219.98167 \mathrm{~mm}) \cdot(50.00148 \mathrm{~mm}) \cdot(60.00175 \mathrm{~mm})-(220 \mathrm{~mm})(50 \mathrm{~mm}) \cdot(60 \mathrm{~mm}) \\
& =659983.503 \mathrm{~mm}^{3}-660000 \mathrm{~mm}^{3}=-16.497 \mathrm{~mm}^{3} .
\end{aligned}
$$

$=$ The reduction in volume.

5- The given solid block has a thickness of 50 mm (in the third dimension, not visible in the figure). Determine the deformation distance $a$ shown in the figure due to the application of the load. Also determine the average normal strain along the diagonal AB , and the average shear strain at corner B . Given $\mathrm{G}=30 \mathrm{GPa}$.

Solution:
The shear stress applied on the block:

$$
\begin{aligned}
\tau & =\frac{30 \times 10^{3} \mathrm{~N}}{(0.05 \mathrm{~m}) \cdot(0.15 \mathrm{~m})} \\
& =4.0 \times 10^{6} \mathrm{~Pa}
\end{aligned}
$$

Using Hooke's Law for shear,


$$
\begin{aligned}
\tau=G, Y \Rightarrow \text { Shear strain, } \begin{aligned}
X_{x 2}=\frac{\tau}{G} & =\frac{4.0 \times 10^{6} \mathrm{~Pa}}{30 \times 10^{9} \mathrm{~Pa}} \\
& =\frac{4.0}{30} \times 10^{-3} \mathrm{rad}
\end{aligned} .
\end{aligned}
$$

Since the shear strain, $\gamma_{x y}$ is the change in angle $\left(\frac{\pi}{2}-\theta^{\circ}\right)$,
then, $\gamma_{x y}=\tan ^{-1}\left(\frac{a}{150 \mathrm{~mm}}\right) \cong \frac{a}{150 \mathrm{~mm}}$ (small strain analysis)

$$
\Rightarrow a=\left(\frac{4.0 \times 10^{-3}}{30} \mathrm{rad}\right) \cdot(150 \mathrm{~mm})=0.02 \mathrm{~mm}
$$

The shear strain © corner $A=$ shear strain © corner $B$

$$
=+\frac{4}{30} \times 10^{-3} \mathrm{rad}=0,00013333 \mathrm{rad}
$$

The normal strain along the diagonal $A B$ :

$$
\begin{aligned}
L_{A B} & =\sqrt{(150 \mathrm{~mm})^{2}+(150 \mathrm{~mm})^{2}}=212.132 \mathrm{~mm} \text { (before deformation). } \\
& L_{A^{\prime} B^{\prime}}=\sqrt{(150 \mathrm{~mm}+0.02 \mathrm{~mm})+(150 \mathrm{~mm})^{2}}=212.146 \mathrm{~mm} \text { (after deformation). } \\
\Rightarrow & \epsilon_{A B}=\frac{L_{A^{\prime} B}-L_{A B}}{L_{A B}}=\frac{212.146-212.132}{212.132}=+6.6 \times 10^{-5} \frac{\mathrm{~mm}}{\mathrm{~mm}}
\end{aligned}
$$

