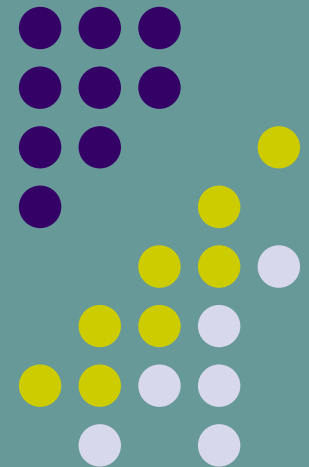


**Minimizing  
and  
maximizing  
compressor and  
turbine work  
respectively**



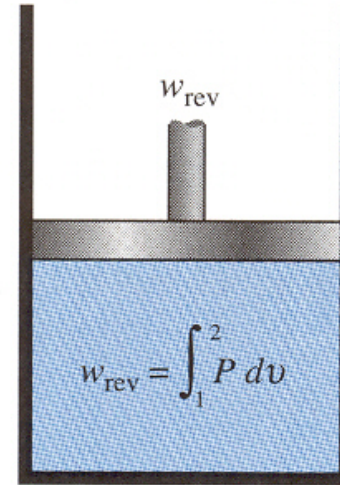
# Reversible steady-flow work

- In Chapter 3, Work Done during a Process was found to be

$$W_b = \int_1^2 P dv$$

- It depends on the path of the process as well as the properties at the end states.

## Work Done during a Process



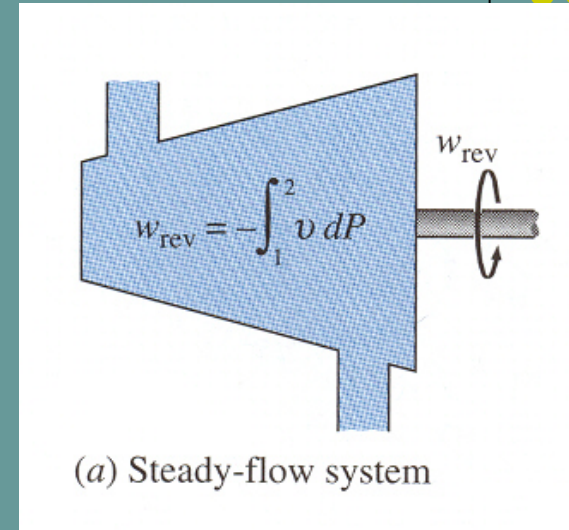
(b) Closed system



# Work Done During a steady state process



In a steady state process, usually there are no moving boundaries



- It would be useful to be able to express the work done during a steady flow process, in terms of system properties
- Recall that steady flow systems work best when they have no irreversibilities

Consider general form of the Energy Balance for steady flow steady state processes

$$\dot{Q} - \dot{W} + \dot{m}_i \left[ h_i + \frac{\vec{V}_i^2}{2} + g(z_i) \right] = \dot{m}_e \left[ h_e + \frac{\vec{V}_e^2}{2} + g(z_e) \right]$$

$$\delta q_{rev} - \delta w_{rev} = dh + dke + dpe$$

$$w_{rev} = - \int_1^2 v dP - \Delta ke - \Delta pe$$

For devices dealing with compressible fluids, like turbines and compressors,  $v$  is not constant, but the KE and PE are negligible. Hence

$$w_{rev} = -\int_1^2 v dP - \cancel{\Delta ke} - \cancel{\Delta pe}$$

$$w_{rev} = -\int_1^2 v dP$$

In order to integrate, we need to know the relationship between  $v$  and  $P$ .

# Important observation



Note that the work term is smallest when  $v$  is small, so for a pump (which uses work) you want  $v$  to be **small**. For a turbine (which produces work) you want  $v$  to be **large**.

$$w_{rev} = - \int_1^2 v dP$$

# Minimizing the Compressor Work



- The best way, is to keep the specific volume as low as possible during the compression process, by **cooling** it

# Maximizing the turbine Work

- The best way, is to keep the specific volume as high as possible during the expansion process, by **heating** it

# Effect of cooling the compressor



- To understand how the cooling affects the work, let us consider three processes:
  - ❖ Isentropic process (No cooling)
  - ❖ Polytropic process (some cooling)
  - ❖ Isothermal process (maximum cooling)
- Assume also that all three processes
  - ❖ Have the same inlet and exit pressures.
  - ❖ Are internally reversible
  - ❖ The gas behaves as an ideal gas
  - ❖ Specific heats are constants.





# 1- Isothermal process

$$W_{rev,in} = \int_1^2 v dP$$

Consider an ideal gas, at constant T

$$v = \frac{RT}{P}$$

$$W_{rev,in} = RT \ln\left(\frac{P_2}{P_1}\right)$$

Remember, this is only true for the isothermal case, for an ideal gas

## 2- Isentropic process



Isentropic means reversible and adiabatic ( $Q=0$ ) i.e. No cooling is allowed

Recall from isentropic relations for an ideal gas

$$Pv^k = C$$

$$v = C^{1/k} P^{-1/k}$$

plug in and integrate

$$w_{rev,in} = \int_1^2 v dP$$

$$w_{rev,in} = \frac{kRT_1}{k-1} \left[ \left( \frac{P_2}{P_1} \right)^{(k-1)/k} - 1 \right]$$

**Remember, this equation only applies to the isentropic case, for an ideal gas, assuming constant specific heats**



# 3- Polytropic process

$$w_{rev,in} = \int_1^2 v dP$$

$$Pv^n = C$$

Back in Chapter 3 we said that in a polytropic process  $Pv^n$  is a constant

This is exactly the same as the isentropic case, but with  $n$  instead of  $k$ !!

$$w_{rev,in} = \frac{v_2 P_2 - v_1 P_1}{1 - 1/n} = \frac{R(T_2 - T_1)}{1 - 1/n} = \frac{nR(T_2 - T_1)}{n - 1}$$

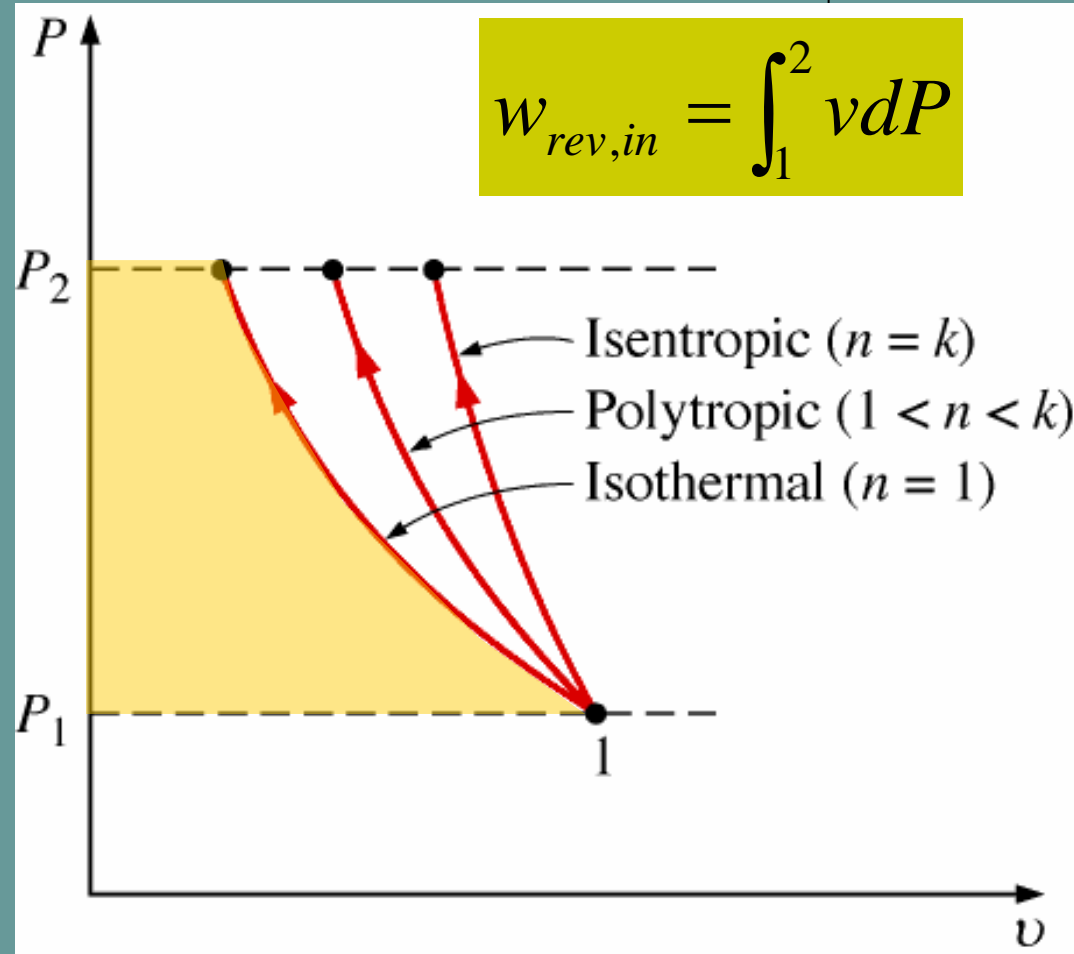
$$w_{rev,in} = \frac{nRT_1}{n - 1} \left[ \left( \frac{P_2}{P_1} \right)^{(n-1)/n} - 1 \right]$$



Let us plot the three processes on a P-v Diagram for the same final and initial pressures

The area to the left of each line represents the work,  $\int v dP$

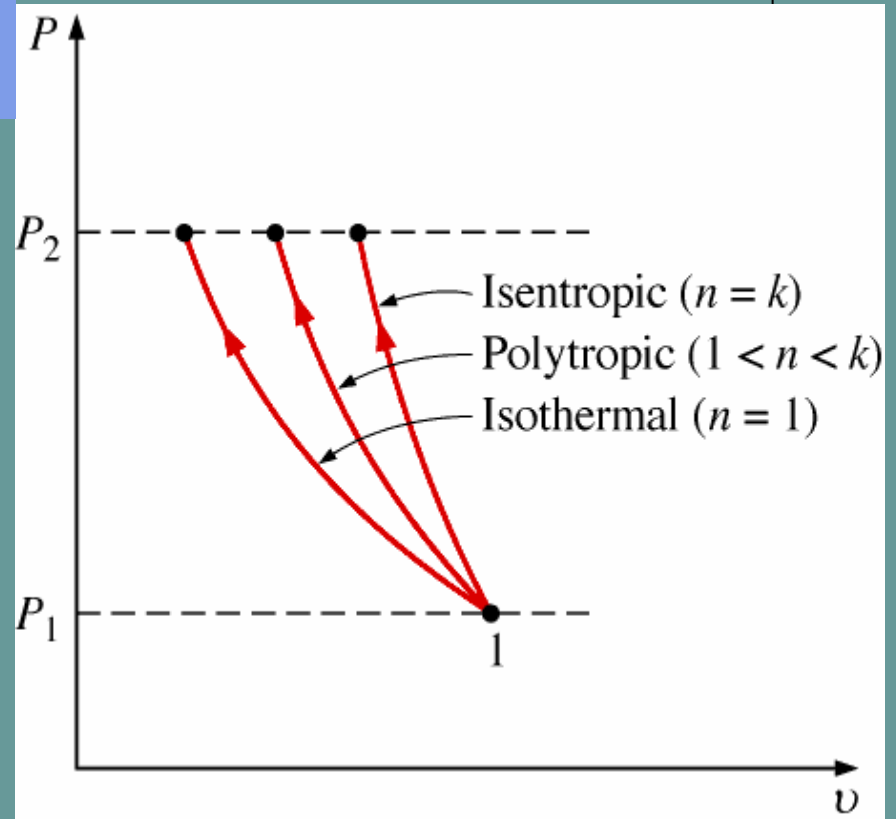
Note, that it takes the maximum work in isentropic compression while it takes minimum work for an isothermal compression





So as an engineer, you should **compress gas isothermally**, in order to consume minimum work.

However, for a turbine, we need to produce the maximum work. So, a turbine should expand **isentropically** (adiabatically and reversibly). That is why we assume  $Q = 0$  in the 1st law analysis of a turbine.



# Multistage compression with intercooling

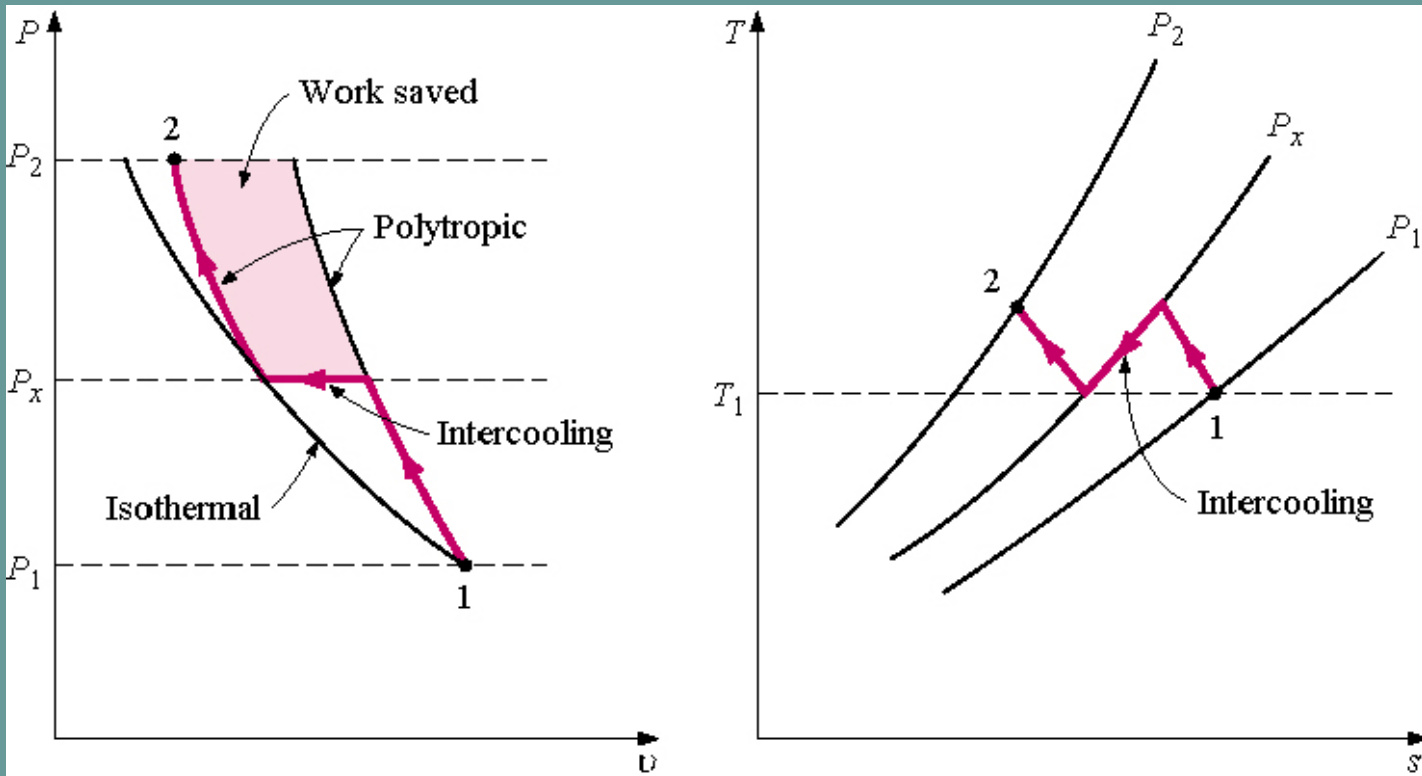


- One common way is to use cooling jackets around the casing of the compressor.
- However, this is not sufficient in some cases.
- Instead, multistage compression is more common, with cooling between steps.
- The gas is compressed in stages and cooled to the initial temperature after each stage.
- This is done by passing it a heat exchanger called “intercooler”.
- Multistage cooling is attractive in high pressure ratio compression.

# Two stage Compressor



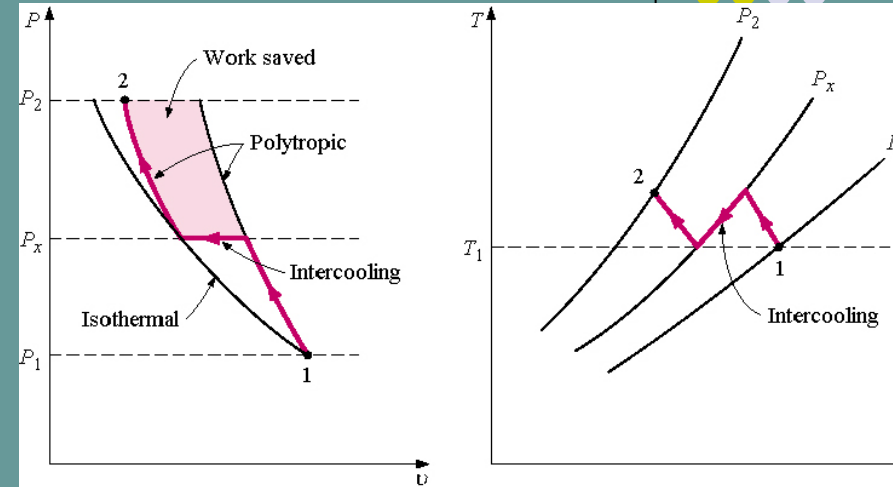
The colored area on the  $P-v$  diagram represents the work saved as a result of two-stage compression with intercooling.



# Minimizing the work input for a two stage Compressor

The size of the colored area (the saved work input) on previous slide varies with the value of the intermediate pressure  $P_x$ .

The total work input for a two-stage compressor is the sum of the work inputs for each stage of compression.



$$\begin{aligned}
 W_{comp,in} &= W_{comp I,in} + W_{comp II,in} \\
 &= \frac{nRT_1}{n-1} \left[ \left( \frac{P_x}{P_1} \right)^{(n-1)/n} - 1 \right] + \frac{nRT_1}{n-1} \left[ \left( \frac{P_2}{P_x} \right)^{(n-1)/n} - 1 \right]
 \end{aligned}$$



The only variable is  $P_x$ .

The  $P_x$  value that will minimize the total work is determined by differentiating the above expression with respect to  $P_x$ . And setting the result to zero.

This gives

$$\left( \frac{P_x}{P_1} \right) = \left( \frac{P_2}{P_x} \right)$$

That is to minimize the compression work during two stage compression, the pressure ratio across each stage of the compressor must be the same.

$$W_{comp I, in} = W_{comp II, in}$$

