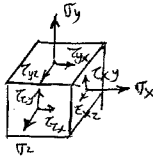


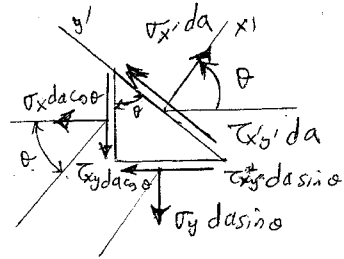
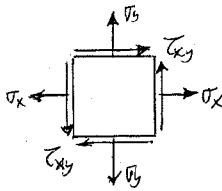
Chapter 7.

Stress Transformation & Deflections

Two - Dimensional Transformation of stress



$$\sigma = \begin{bmatrix} \sigma_x & \tau_{xy} & \tau_{xz} \\ \tau_{yx} & \sigma_y & \tau_{yz} \\ \tau_{zx} & \tau_{zy} & \sigma_z \end{bmatrix} \begin{array}{l} \leftarrow x\text{-plane} \\ \leftarrow y\text{-plane} \\ \leftarrow z\text{-plane} \end{array}$$



$$\Sigma F_{x'} = 0: \sigma'_x da - \tau_{xy} da \sin \theta \cos \theta - \tau_{yx} da \cos \theta \sin \theta - \sigma_x da \cos \theta \cos \theta - \sigma_y da \sin \theta \sin \theta = 0$$

$$\sigma'_x - \sigma_x \cos^2 \theta - \sigma_y \sin^2 \theta - 2\tau_{xy} \sin \theta \cos \theta = 0$$

$$\Rightarrow \sigma'_x = \sigma_x \cos^2 \theta + \sigma_y \sin^2 \theta + 2\tau_{xy} \sin \theta \cos \theta \quad \text{--- (1)}$$

$$\Sigma F_{y'} = 0: \tau'_{xy} = -(\sigma_x - \sigma_y) \sin \theta \cos \theta + \tau_{xy} (\cos^2 \theta - \sin^2 \theta) \quad \text{--- (2)}$$

Using the following info (1) & (2) \Rightarrow

$$\left. \begin{array}{l} \cos^2 \theta = \frac{1}{2}(1 + \cos 2\theta) \\ \sin^2 \theta = \frac{1}{2}(1 - \cos 2\theta) \\ \sin \theta \cos \theta = \frac{1}{2} \sin 2\theta \\ \cos^2 \theta - \sin^2 \theta = \cos 2\theta \end{array} \right\} \Rightarrow$$

$$\sigma'_x = \frac{\sigma_x + \sigma_y}{2} + \frac{\sigma_x - \sigma_y}{2} \cos 2\theta + \tau_{xy} \sin 2\theta \quad \text{--- (A)}$$

$$\tau'_{xy} = -\left(\frac{\sigma_x - \sigma_y}{2}\right) \sin 2\theta + \tau_{xy} \cos 2\theta \quad \text{--- (B)}$$

to find σ'_y , use (A) with $\theta = \theta + \pi/2 \Rightarrow$

$$\sigma'_y = \frac{\sigma_x + \sigma_y}{2} - \left(\frac{\sigma_x - \sigma_y}{2}\right) \cos 2\theta - \tau_{xy} \sin 2\theta \quad \text{--- (C)}$$

Also $\sigma_x + \sigma_y = \sigma'_x + \sigma'_y$

1m²

 20kN
 $\sigma_{all} = 25 \text{ kPa}$
 $\sigma_{cu} = 5 \text{ kPa}$
 Is it safe

Principal Values and Principal Directions

To find the maximum and minimum values of the normal stress $\sigma_x' = \sigma_x'(\sigma_x, \sigma_y, \tau_{xy}, \theta)$, one needs

to set $\frac{d\sigma_x'}{d\theta} = 0$ and find θ_p then substitute to find σ_{\max} and σ_{\min} .

$$\frac{d\sigma_x'}{d\theta} = 0 \Rightarrow \tan 2\theta_p = \frac{\tau_{xy}}{\left(\frac{\sigma_x - \sigma_y}{2}\right)}$$

which upon substitution into (A), gives

$$\begin{cases} \sigma_{\max} \\ \sigma_{\min} \end{cases} = \frac{\sigma_x + \sigma_y}{2} \pm \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2}$$

which from (B) gives $\tau = 0$

\therefore Planes of principal normal stress have no shear stress

Similarly to find maximum shear stress set

$$\frac{d\tau_{xy}'}{d\theta} = 0 \Rightarrow \tan 2\theta_s = \frac{-\left(\frac{\sigma_x - \sigma_y}{2}\right)}{\tau_{xy}}$$

which upon substitution into (B) gives

$$\begin{cases} \tau_{\max} \\ \tau_{\min} \end{cases} = \pm \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2}$$

which upon substitution into (A) gives

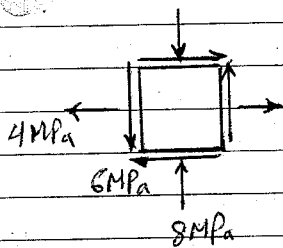
$\sigma = \frac{\sigma_x + \sigma_y}{2}$, the corresponding normal stress on planes of principal shear stress.

Given the state of stress at a material point as shown below, find the following in two ways: using equations and Mohr's Circle.

1] Find the principal normal stress (magnitude + direction) show them on properly oriented element with respect to x-y axes.

2] Find the principal shear stress (magnitude + direction) show them on properly oriented element. DO NOT FORGET the normal stresses that are acting with shear.

3] Find the stress state (normal and shear stresses) at plane making an angle $+30^\circ$.



$$\sigma_x = 4 \text{ MPa}$$

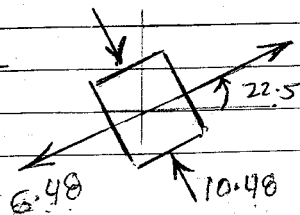
$$\sigma_y = -8 \text{ MPa}$$

$$\tau_{xy} = 6 \text{ MPa}$$

Equation Method:

$$\sigma_{\max/\min} = \frac{\sigma_x + \sigma_y}{2} \pm \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2} \Rightarrow \sigma_{\max/\min} = \begin{cases} 6.48 \\ -10.48 \end{cases}$$

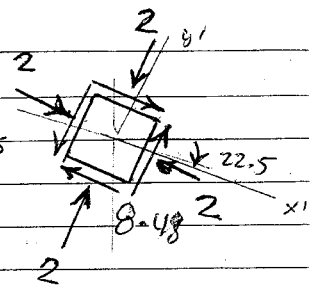
$$\tan 2\theta_p = \frac{\tau_{xy}}{\left(\frac{\sigma_x - \sigma_y}{2}\right)} \Rightarrow \theta_p = 22.5^\circ$$



2) $\tau_{max/min} = \pm \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2} = \pm 8.48$

$\tan 2\theta_s = \frac{-(\sigma_x - \sigma_y)/2}{\tau_{xy}} \Rightarrow \theta_s = -22.5$

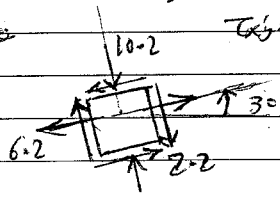
$\tau_{avg} = \frac{\sigma_x + \sigma_y}{2} = -2$



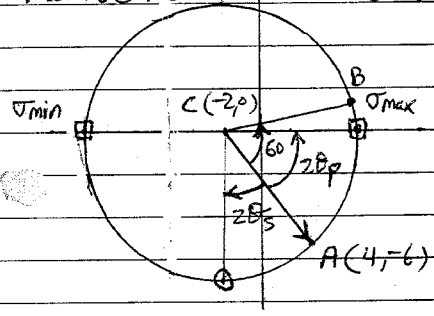
3) $\sigma_{x'} = \frac{\sigma_x + \sigma_y}{2} + \frac{\sigma_x - \sigma_y}{2} \cos 2\theta + \tau_{xy} \sin 2\theta \Rightarrow \sigma_{x'} = 6.2$

$\tau_{x'y'} = \frac{\sigma_x - \sigma_y}{2} \sin 2\theta + \tau_{xy} \cos 2\theta \Rightarrow \tau_{x'y'} = -2.2$

$\sigma_{y'} = \sigma_x + \sigma_y - \sigma_{x'} = -10.2$



Mohr's circle Method:



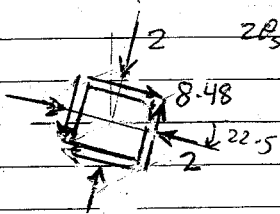
$R = \text{distance between } C \text{ and } A$
 $= \sqrt{(-2-4)^2 + 6^2} = 8.48$

1) $\sigma_{max} = C + R = 6.48$

$\sigma_{min} = C - R = -10.48$

$\sin 2\theta_p = \frac{6}{8.48} \Rightarrow \theta_p = 22.5^\circ$

2) $\tau_{max/min} = \pm R = \pm 8.48$



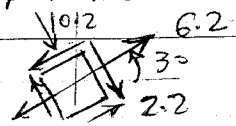
$2\theta_s = 90 - 2\theta_p \Rightarrow \theta_s = 22.5^\circ$

3) In order to find the stress at a plane +30 degrees, you need to go from A counterclockwise 60, which gives point B.

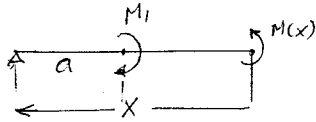
Then stress at the plane nothing but the coordinate of B:

$\sigma = R \cos 15 - 2 = 6.2$

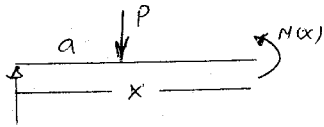
$\tau = R \sin 15 = +2.2$



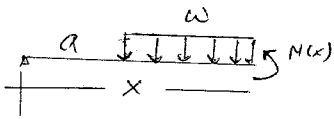
Singularities Function:



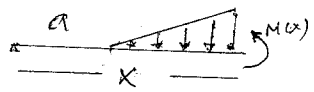
$$M(x) = +M_1 \langle x-a \rangle^0$$



$$M(x) = -P \langle x-a \rangle^1$$



$$M(x) = -\frac{w}{2} \langle x-a \rangle^2$$

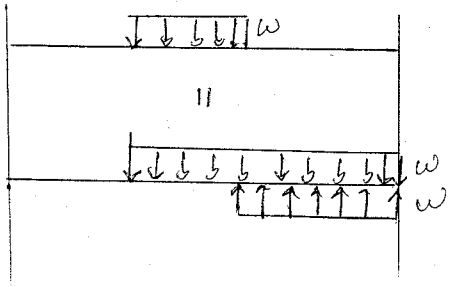


$$M(x) = -\frac{k}{6} \langle x-a \rangle^3$$

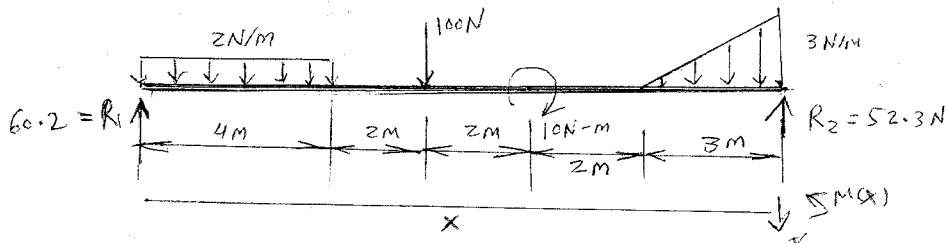
Macaulay brackets : $\langle +x \rangle = +x$
 $\langle -x \rangle = 0$

Remember distributed loads can start anywhere but should continue to the end of the beam; otherwise it should be forced to go to end.

Ex.



Example: Find the slope of the beam at support A



$$EI v''(x) = 60.2 \langle x-0 \rangle - \frac{2}{2} \langle x-0 \rangle^2 + \frac{2}{2} \langle x-4 \rangle^2 - 100 \langle x-6 \rangle + 10 \langle x-8 \rangle^0 - \frac{3/3}{6} \langle x-10 \rangle^3$$

$$EI v'(x) = \frac{60.2}{2} \langle x-0 \rangle^2 - \frac{1}{3} \langle x-0 \rangle^3 + \frac{1}{3} \langle x-4 \rangle^3 - 50 \langle x-6 \rangle^2 + 10 \langle x-8 \rangle - \frac{1}{24} \langle x-10 \rangle^4 + C_1$$

$$EI v(x) = \frac{60.2}{6} \langle x-0 \rangle^3 - \frac{1}{12} \langle x-0 \rangle^4 + \frac{1}{12} \langle x-4 \rangle^4 - \frac{50}{3} \langle x-6 \rangle^3 + \frac{10}{2} \langle x-8 \rangle^2 - \frac{1}{24 \times 5} \langle x-10 \rangle^5 + C_1 \langle x-0 \rangle + C_2$$

$v(0) = 0 \Rightarrow C_2 = 0$

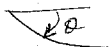
$v(13) = 0 \Rightarrow C_1 = -1124$

$$v(x) = \frac{1}{EI} \left\{ \frac{60.2}{6} \langle x-0 \rangle^3 - \frac{1}{12} \langle x-0 \rangle^4 + \frac{1}{12} \langle x-4 \rangle^4 - \frac{50}{3} \langle x-6 \rangle^3 + 5 \langle x-8 \rangle^2 - \frac{1}{120} \langle x-10 \rangle^5 - 1124 \langle x-0 \rangle \right\}$$

$v'(x=0) = \text{slope at support A}$

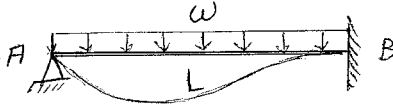
$= C_1 / EI$

$= -1124 / EI$



Using Singularity Functions to Determine Reactions of Statically Indeterminate Problems:

Example ①: Find reaction at support A



$$EI v''(x) = M(x)$$

$$EI v''(x) = A_y \langle x-0 \rangle - \frac{w}{2} \langle x-0 \rangle^2$$

$$EI v'(x) = \frac{A_y}{2} \langle x-0 \rangle^2 - \frac{w}{6} \langle x-0 \rangle^3 + C_1$$

$$EI v(x) = \left(\frac{A_y}{6}\right) \langle x-0 \rangle^3 - \frac{w}{24} \langle x-0 \rangle^4 + C_1 \langle x-0 \rangle + C_2$$

unknowns: $\{ C_1, C_2, A_y \}$

Boundary Conditions:

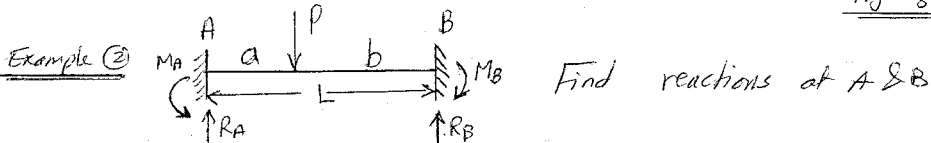
$$\begin{cases} v(0) = 0 & \Rightarrow C_2 = 0 & \text{--- (1)} \\ v(L) = 0 & \Rightarrow \frac{A_y L^3}{6} - \frac{wL^4}{24} + C_1 L = 0 & \text{--- (2)} \\ v'(L) = 0 & \Rightarrow \frac{A_y L^2}{2} - \frac{wL^3}{6} + C_1 = 0 & \text{--- (3)} \end{cases}$$

Solving (1) - (3) \Rightarrow

$$C_1 = -wL^3/48$$

$$C_2 = 0$$

$$\underline{A_y = \frac{3}{8} wL}$$



$$EI v''(x) = R_A \langle x-0 \rangle - M_A \langle x-0 \rangle - P \langle x-a \rangle$$

$$EI v'(x) = \frac{R_A}{2} \langle x-0 \rangle^2 - M_A \langle x-0 \rangle - \frac{P}{2} \langle x-a \rangle^2 + C_1$$

$$EI v(x) = \left(\frac{R_A}{6}\right) \langle x-0 \rangle^3 - \left(\frac{M_A}{2}\right) \langle x-0 \rangle^2 - \left(\frac{P}{6}\right) \langle x-a \rangle^3 + C_1 \langle x-0 \rangle + C_2$$

unknowns: $\{ C_1, C_2, R_A, M_A \}$

Boundary conditions: $\{ v(0) = 0; v'(0) = 0; v(L) = 0; v'(L) = 0 \}$

$$v(0) = 0 : C_2 = 0$$

$$v'(0) = 0 : C_1 = 0$$

$$\left. \begin{aligned} v(L) = 0 : \frac{R_A}{6} L^3 - \frac{M_A}{2} L^2 - \frac{P}{6} b^3 = 0 \\ v'(L) = 0 : \frac{R_A}{2} L^2 - M_A L - \frac{P}{2} b^2 = 0 \end{aligned} \right\} \Rightarrow \begin{aligned} R_A &= \frac{Pb^2}{L^3} (3L - 2b) \\ M_A &= Pab^2/L^2 \end{aligned}$$

one can use equilibrium equations to find the remaining unknowns (R_B, M_B)